



# ***Aerospace Measurement and Experimental System Development Characterization***

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# Overview



- **Aerospace measurement and experimental system development characterization for research and development presents opportunities for**
  - **innovative applications of existing statistical methods**
  - **impetus for statistical research**

## Highlighted Methods

- **Inverse Regression**
- **Response Surface Methodology for Characterization**
- **Iterative, Inverse Prediction and Prediction Intervals**
- **Process optimization**

# Aerospace R&D Characterization vs. Classical Calibration



<b>Aerospace R&amp;D</b>	<b>Calibration</b>
One-of-a-kind, application specific measurement system	Common, off-the-shelf instrument
Tested in a unique environmental simulation facility, used in flight or flight-like conditions	Tested in laboratory controlled conditions, used in secondary controlled laboratory
Known, traceable standards are often not available	Physical, NIST traceable standards
Multiple-sensing device	Element, measuring one property
Multi-dimensional response surface	Simple linear regression
Inverse regression broadly used by engineers due to simplicity	Classical regression, inverse solution

# Simple Linear, Single Measurement

## 1 Factor, 1 Response, 1<sup>st</sup>-Order



- Airborne Subscale Transport Aircraft Research (AirSTAR)
- Dynamically scaled, commercial aircraft to
  - study control-upset conditions
  - improve pilot training

- Pressure measurement system for altitude and airspeed on wing tips of vehicle
- System testing performed in laboratory, used in open-air flight

$$V = b_0 + b_1P + e$$



# ***Classical and Inverse Regression***

## ***1 Factor, 1 Response, 1<sup>st</sup>-Order***



### **Classical Regression Model**

$$y = b_0 + b_1x + e$$

### **Inverse Application**

$$\hat{x} = \frac{y - \hat{\beta}_0}{\hat{\beta}_1}$$

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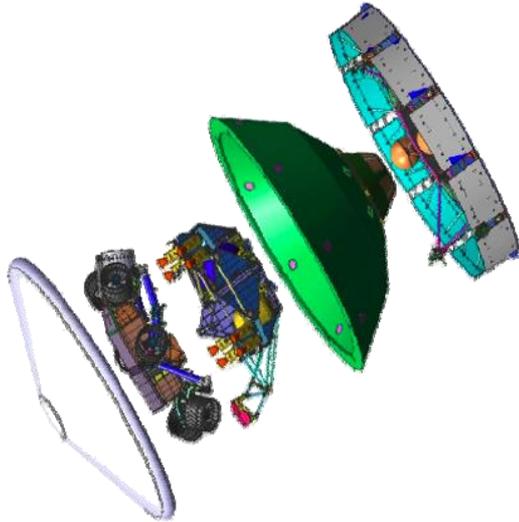
### **Inverse Regression Model**

$$x = g_0 + g_1y + e$$

- **Reversing the role of the  $x$  and  $y$  is commonly done in practice**
- **Both approaches lead to biased predictions, classical is slightly better**
- **Prediction intervals have essentially correct coverage probabilities**
- **Inverse interval width is slightly smaller and less variable**

# Multi-dimensional Response Surface

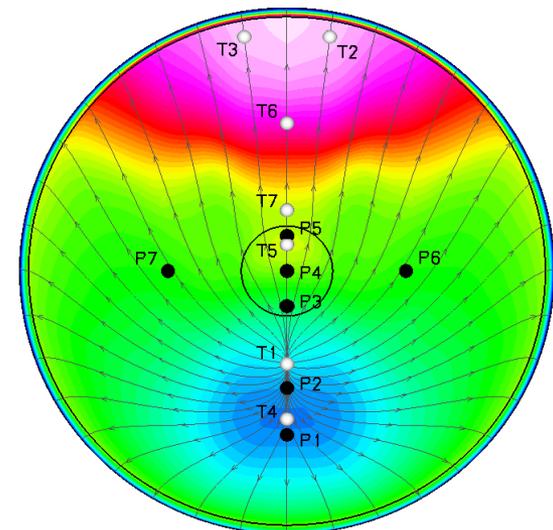
3 Factors, 1 Response, 2<sup>nd</sup>-Order



- Reduce uncertainty in pre-flight landing ellipse estimation through measurements during Mars entry
- Pressure measurements during extreme atmospheric entry temperature conditions

- Pressure measurement system is sensitive to temperature
- Signal ( $V$ ) as a function of pressure and two temperatures
- Second-order Response Surface

$$V = f(P, T_1, T_2) + e$$



# Response Surface for Characterization

3 Factors, 1 Response, 2<sup>nd</sup>-Order



- Response Surface Methods for a non-traditional application
  - Characterization, not optimization
  - Efficiency in achieving absolute prediction variance, not per point
  - Mathematical model delivered, not optimized factor settings
  - Confirmation points to test the model over the entire design space, not sensitivity to the location of optimum performance
- Inverse Prediction of Second-Order Response Surface
  - Iterative procedure employed, (direct, quadratic formula issues)

$$\hat{P} = \hat{f}(V, T_1, T_2) = f\left(\hat{b}_0, V, T_1, T_2, \hat{b}_1, \hat{b}_2, \hat{b}_3, \hat{b}_{11}, \hat{b}_{12}, \hat{b}_{22}\right)$$

- Approximate inverse prediction intervals from the Delta Method

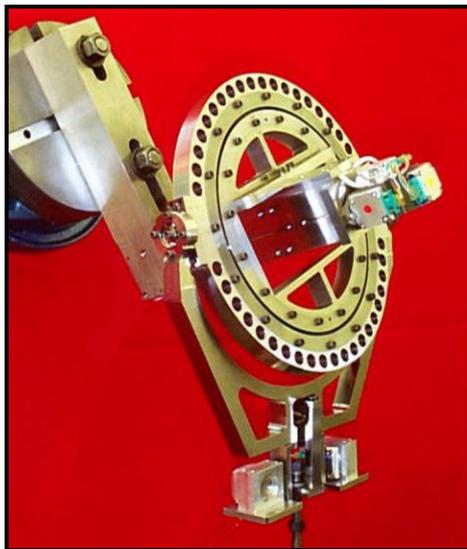
$$\left(\hat{P}(\hat{\mathbf{x}})\right)^T \left(\mathbf{X}'\hat{\mathbf{S}}^{-1}\mathbf{X}\right)^{-1} \left(\hat{P}(\hat{\mathbf{x}})\right)$$

# Multi-input, Multi-output, Higher Order

## 6 Factors, 6 Responses, 2<sup>nd</sup> and higher



- Multi-component force transducers used in aerospace research and development
- Sensing 3 forces and 3 moments, simultaneously
- No system calibration standards



- Modeling

$$\mathring{a} \underset{i=1}{y}_i = f_i \left( [x_1 \cdots x_6] \right) + e_i$$

- Inverse prediction

$$[\hat{x}_1 \cdots \hat{x}_6] = \hat{F} \left( [y_1 \cdots y_6] \right)$$

# Multivariable Response Surface

## 6 Factors, 6 Responses, 2<sup>nd</sup> and higher



- Internationally, some use inverse regression
  - simplified, direct solution – properties are not well-defined
- Multivariate version of Delta Method, inverse prediction intervals

$$\left[ \text{var}(\hat{x}_1) \cdots \text{var}(\hat{x}_6) \right]^T = \left[ \text{var}(\hat{y}_1(\hat{\mathbf{x}})) \cdots \text{var}(\hat{y}_6(\hat{\mathbf{x}})) \right]^T \frac{\partial \hat{\mathbf{F}}(\hat{\mathbf{x}})}{\partial \hat{\mathbf{x}}} \hat{\mathbf{U}}^{-1}$$

Response surfaces of cubic or higher are feasible

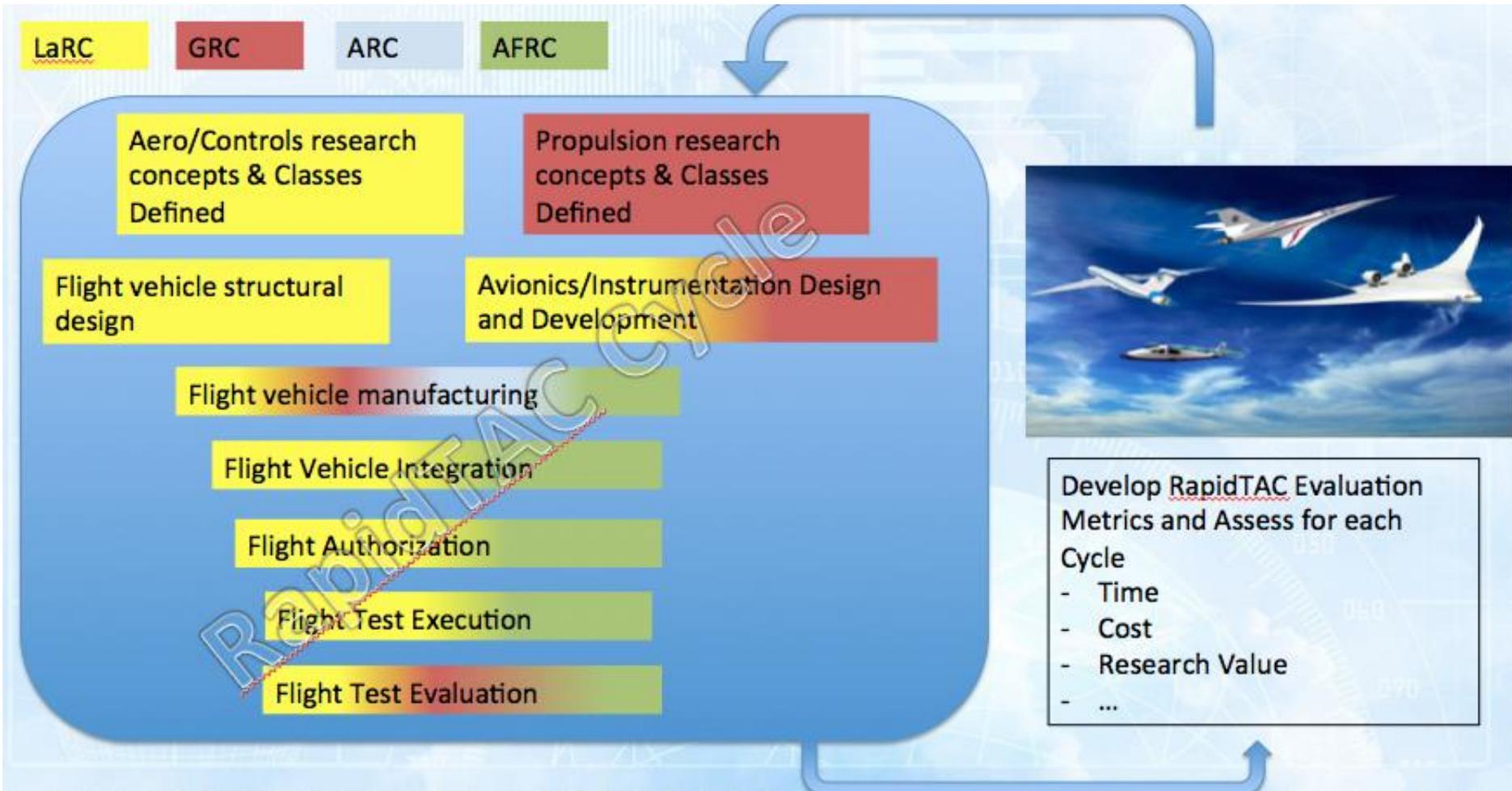
- Cubic designs based on combining two second-order designs
- Design, modeling, inverse prediction extended to higher order

# Experimental System Development



## Example: Rapid Test of Aeronautics Concepts (RapidTAC)

- Process optimization with unique experiments
- Quantification of research value and complexity



# ***Concluding Remarks***



- **Aerospace measurement and experimental system development characterization for research and development**
  - similar to classical calibration in concept
  - requires adaption and extension of existing statistical methods
- **Methods highlighted**
  - Inverse Regression
  - RSM for Characterization
  - Inverse Prediction and Intervals
  - Process optimization