DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2020-0092; Product Identifier 2020-NM-001-AD; Amendment 39-19905; AD 2020-08-13]

RIN 2120-AA64

Airworthiness Directives; Bombardier, Inc. Airplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: The FAA is adopting a new airworthiness directive (AD) for certain Bombardier, Inc., Model CL-600-2B19 (Regional Jet Series 100 & 440) airplanes, Model CL-600-2C10 (Regional Jet Series 700, 701 & 702) airplanes, Model CL-600-2D15 (Regional Jet Series 705) airplanes, and Model CL-600-2D24 (Regional Jet Series 900) airplanes; and all Bombardier, Inc., Model CL-600-2C11 (Regional Jet Series 550) airplanes. This AD was prompted by reports of fractured rudder primary feel unit shafts; a subsequent investigation determined that the fractures in the shafts are consistent with fatigue damage. This AD requires replacement of the rudder primary feel unit shaft. The FAA is issuing this AD to address the unsafe condition on these products.

DATES: This AD is effective [INSERT DATE 35 DAYS AFTER DATE OF PUBLICATION IN THE FEDERAL REGISTER].
The Director of the Federal Register approved the incorporation by reference of certain publications listed in this AD as of [INSERT DATE 35 DAYS AFTER DATE OF PUBLICATION IN THE FEDERAL REGISTER].

**ADDRESSES:** For service information identified in this final rule, contact Bombardier, Inc., 400 Côte-Vertu Road West, Dorval, Québec H4S 1Y9, Canada; Widebody Customer Response Center North America toll-free telephone 1-866-538-1247 or direct-dial telephone 1-514-855-2999; fax 514-855-7401; email ac.yul@aero.bombardier.com; Internet http://www.bombardier.com. You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195. It is also available on the Internet at https://www.regulations.gov by searching for and locating Docket No. FAA-2020-0092.

**Examining the AD Docket**

You may examine the AD docket on the Internet at https://www.regulations.gov by searching for and locating Docket No. FAA-2020-0092; or in person at Docket Operations between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains this final rule, any comments received, and other information. The address for Docket Operations is U.S. Department of Transportation, Docket Operations, M-30, West Building Ground Floor, Room W12-140, 1200 New Jersey Avenue SE., Washington, DC 20590.

**FOR FURTHER INFORMATION CONTACT:** Andrea Jimenez, Aerospace Engineer, Airframe and Mechanical Systems Section, FAA, New York ACO Branch,
TRANSPORT CANADA CIVIL AVIATION (TCCA), which is the aviation authority for Canada, has issued Canadian AD CF-2019-42, dated November 8, 2019 (also referred to as the Mandatory Continuing Airworthiness Information, or “the MCAI”), to correct an unsafe condition for certain Bombardier, Inc., Model CL-600-2B19 (Regional Jet Series 100 & 440) airplanes, Model CL-600-2C10 (Regional Jet Series 700, 701 & 702) airplanes, Model CL-600-2D15 (Regional Jet Series 705) airplanes, and Model CL-600-2D24 (Regional Jet Series 900) airplanes; and all Bombardier, Inc., Model CL-600-2C11 (Regional Jet Series 550) airplanes. You may examine the MCAI in the AD docket on the Internet at https://www.regulations.gov by searching for and locating Docket No. FAA-2020-0092.

The FAA issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 by adding an AD that would apply to certain Bombardier, Inc., Model CL-600-2B19 (Regional Jet Series 100 & 440) airplanes, Model CL-600-2C10 (Regional Jet Series 700, 701 & 702) airplanes, Model CL-600-2D15 (Regional Jet Series 705) airplanes, and Model CL-600-2D24 (Regional Jet Series 900) airplanes; and all Bombardier, Inc., Model CL-600-2C11 (Regional Jet Series 550) airplanes. The NPRM published in the Federal Register on February 18, 2020 (85 FR 8768). The NPRM was prompted by reports of fractured rudder primary feel unit shafts; a subsequent investigation determined that the fractures in the shafts are consistent with fatigue damage. The NPRM proposed
to require replacement of the rudder primary feel unit shaft. The FAA is issuing this AD to address fractures of the rudder primary feel unit shaft, which could result in a loss of feel in the yaw axis and thereby impact the controllability of the airplane. See the MCAI for additional background information.

Comments

The FAA gave the public the opportunity to participate in developing this final rule. The FAA has considered the comments received. The Air Line Pilots Association, International (ALPA) and Jacob Yepiz stated support for the NPRM.

Conclusion

The FAA reviewed the relevant data, considered the comments received, and determined that air safety and the public interest require adopting this final rule as proposed, except for minor editorial changes. The FAA has determined that these minor changes:

- Are consistent with the intent that was proposed in the NPRM for addressing the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM.

Related Service Information under 1 CFR Part 51

Bombardier has issued Service Bulletin 601R-27-166, dated April 5, 2019; and Service Bulletin 670BA-27-075, dated April 5, 2019. This service information describes procedures for replacing the rudder primary feel unit shaft that has part number 600-90251-1 with a new shaft. These documents are distinct since they apply to different airplane models. This service information is reasonably available because the interested
parties have access to it through their normal course of business or by the means identified in the ADDRESSES section.

Costs of Compliance

The FAA estimates that this AD affects 1,002 airplanes of U.S. registry. The FAA estimates the following costs to comply with this AD:

**Estimated costs for required actions**

<table>
<thead>
<tr>
<th>Labor cost</th>
<th>Parts cost</th>
<th>Cost per product</th>
<th>Cost on U.S. operators</th>
</tr>
</thead>
<tbody>
<tr>
<td>18 work-hours X $85 per hour = $1,530</td>
<td>$158</td>
<td>$1,688</td>
<td>$1,691,376</td>
</tr>
</tbody>
</table>

According to the manufacturer, some or all of the costs of this AD may be covered under warranty, thereby reducing the cost impact on affected individuals. The FAA does not control warranty coverage for affected individuals. As a result, the FAA has included all known costs in our cost estimate.

Authority for this Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. Subtitle VII: Aviation Programs, describes in more detail the scope of the Agency’s authority.

The FAA is issuing this rulemaking under the authority described in Subtitle VII, Part A, Subpart III, Section 44701: “General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority.
because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

**Regulatory Findings**

This AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.

For the reasons discussed above, I certify that this AD:

(1) Is not a “significant regulatory action” under Executive Order 12866,

(2) Will not affect intrastate aviation in Alaska, and

(3) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

**List of Subjects in 14 CFR Part 39**

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

**Adoption of the Amendment**

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

**PART 39 - AIRWORTHINESS DIRECTIVES**

1. The authority citation for part 39 continues to read as follows:

   Authority: 49 U.S.C. 106(g), 40113, 44701.

   § 39.13 [Amended]

   2. The FAA amends § 39.13 by adding the following new airworthiness directive (AD):
2020-08-13 Bombardier, Inc.: Amendment 39-19905; Docket No. FAA-2020-0092; Product Identifier 2020-NM-001-AD.

(a) Effective Date

This AD is effective [INSERT DATE 35 DAYS AFTER DATE OF PUBLICATION IN THE FEDERAL REGISTER].

(b) Affected ADs

None.

(c) Applicability

This AD applies to Bombardier, Inc. airplanes, certificated in any category, as specified in paragraphs (c)(1) through (4) of this AD.

(1) Model CL-600-2B19 (Regional Jet Series 100 & 440) airplanes, serial numbers (S/Ns) 7003 through 7990 inclusive, and S/Ns 8000 and subsequent.

(2) Model CL-600-2C10 (Regional Jet Series 700, 701 & 702) airplanes, S/Ns 10002 through 10347 inclusive.

(3) Model CL-600-2D15 (Regional Jet Series 705) airplanes and Model CL-600-2D24 (Regional Jet Series 900) airplanes, S/Ns 15001 through 15469 inclusive.

(4) Model CL-600-2C11 (Regional Jet Series 550) airplanes, all serial numbers.

(d) Subject

Air Transport Association (ATA) of America Code 27, Flight controls.

(e) Reason

This AD was prompted by reports of fractured rudder primary feel unit shafts; a subsequent investigation determined that the fractures in the shafts are consistent with fatigue damage. The FAA is issuing this AD to address fractures of the rudder primary
feel unit shaft, which could result in a loss of feel in the yaw axis and thereby impact the controllability of the airplane.

(f) Compliance

Comply with this AD within the compliance times specified, unless already done.

(g) Replacement

Within the compliance times specified in figure (1) to paragraph (g) of this AD:

Replace all rudder primary feel unit shafts that have part number (P/N) 600-90251-1 with a new shaft, in accordance with the Accomplishment Instructions of the Bombardier Service Bulletin 601R-27-166, dated April 5, 2019; or Bombardier Service Bulletin 670BA-27-075, dated April 5, 2019; as applicable. For Model CL-600-2C11 (Regional Jet Series 550) airplanes, do the replacement in accordance with the Accomplishment Instructions of Bombardier Service Bulletin 670BA-27-075, dated April 5, 2019.

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**Figure 1 to paragraph (g) – Compliance Times**

<table>
<thead>
<tr>
<th>Accumulated Airplane Flight Cycles</th>
<th>Compliance Time for the Replacement</th>
</tr>
</thead>
<tbody>
<tr>
<td>For airplanes that have accumulated 22,000 total flight cycles or less as of the effective date of this AD.</td>
<td>Before the airplane reaches 30,000 total flight cycles.</td>
</tr>
<tr>
<td>For airplanes that have accumulated more than 22,000 total flight cycles, but less than 37,000 total flight cycles, as of the effective date of this AD.</td>
<td>Within 8,000 flight cycles from the effective date of this AD.</td>
</tr>
<tr>
<td>For airplanes that have accumulated 37,000 total flight cycles or more, but less than 40,000 total flight cycles, as of the effective date of this AD.</td>
<td>Before the airplane reaches 45,000 total flight cycles.</td>
</tr>
<tr>
<td>For airplanes that have accumulated 40,000 total flight cycles or more, but less than 46,500 total flight cycles, as of the effective date of this AD.</td>
<td>Within 5,000 flight cycles from the effective date of this AD.</td>
</tr>
</tbody>
</table>
## Accumulated Airplane Flight Cycles

<table>
<thead>
<tr>
<th>Compliance Time for the Replacement</th>
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<tbody>
<tr>
<td>For airplanes that have accumulated 46,500 total flight cycles or more as of the effective date of this AD.</td>
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</tbody>
</table>

### (h) Parts Installation Prohibition

As of the effective date of this AD, no person may install a rudder primary feel unit shaft, P/N 600-90251-1, on any airplane.

### (i) Other FAA AD Provisions

The following provisions also apply to this AD:

1. **Alternative Methods of Compliance (AMOCs):** The Manager, New York ACO Branch, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. In accordance with 14 CFR 39.19, send your request to your principal inspector or local Flight Standards District Office, as appropriate. If sending information directly to the manager of the certification office, send it to ATTN: Program Manager, Continuing Operational Safety, FAA, New York ACO Branch, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516-228-7300; fax 516-794-5531. Before using any approved AMOC, notify your appropriate principal inspector, or lacking a principal inspector, the manager of the local flight standards district office/certificate holding district office.

2. **Contacting the Manufacturer:** For any requirement in this AD to obtain instructions from a manufacturer, the instructions must be accomplished using a method approved by the Manager, New York ACO Branch, FAA; or Transport Canada Civil Aviation (TCCA); or Bombardier, Inc.’s TCCA Design Approval Organization (DAO). If approved by the DAO, the approval must include the DAO-authorized signature.
(j) Related Information

(1) Refer to Mandatory Continuing Airworthiness Information (MCAI) Canadian AD CF-2019-42, dated November 8, 2019, for related information. This MCAI may be found in the AD docket on the Internet at https://www.regulations.gov by searching for and locating Docket No. FAA-2020-0092.

(2) For more information about this AD, contact Andrea Jimenez, Aerospace Engineer, Airframe and Mechanical Systems Section, FAA, New York ACO Branch, 1600 Stewart Avenue, Suite 410, Westbury, NY 11590; telephone 516-228-7330; fax 516-794-5531; email 9-avs-nyaco-cos@faa.gov.

(k) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless this AD specifies otherwise.


(3) For service information identified in this AD, contact Bombardier, Inc., 400 Côte-Vertu Road West, Dorval, Québec H4S 1Y9, Canada; Widebody Customer Response Center North America toll-free telephone 1-866-538-1247 or direct-dial telephone 1-514-855-2999; fax 514-855-7401; email ac.yul@aero.bombardier.com; Internet http://www.bombardier.com.
(4) You may view this service information at the FAA, Airworthiness Products Section, Operational Safety Branch, 2200 South 216th St., Des Moines, WA. For information on the availability of this material at the FAA, call 206-231-3195.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the availability of this material at NARA, email fedreg.legal@nara.gov, or go to: https://www.archives.gov/federal-register/cfr/ibr-locations.html.

Issued on April 20, 2020.

Gaetano A. Sciortino, Deputy Director for Strategic Initiatives, Compliance & Airworthiness Division, Aircraft Certification Service.

[FR Doc. 2020-09663 Filed: 5/5/2020 8:45 am; Publication Date: 5/6/2020]