DEPARTMENT OF TRANSPORTATION

Federal Aviation Administration

14 CFR Part 39

[Docket No. FAA-2014-0180; Directorate Identifier 2014-CE-004-AD; Amendment 39-17869; AD 2014-12-08 ]

RIN 2120-AA64

Airworthiness Directives; Przedsiebiorstwo Doswiadczalno-Produkcyjne Szybownictwa “PZL-Bielsko” Model SZD-50-3 “Puchacz” Sailplanes

AGENCY: Federal Aviation Administration (FAA), Department of Transportation (DOT).

ACTION: Final rule.

SUMMARY: We are superseding Airworthiness Directive (AD) 2004-11-10 for Przedsiebiorstwo Doswiadczalno-Produkcyjne Szybownictwa “PZL-Bielsko” Model SZD-50-3 “Puchacz” sailplanes. This AD results from mandatory continuing airworthiness information (MCAI) issued by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as fatigue damage of the welded joint between the airbrake torque tube and the airbrake control system lever located inside the fuselage. We are issuing this AD to require actions to address the unsafe condition on these products.

DATES: This AD is effective [INSERT DATE 35 DAYS AFTER DATE OF PUBLICATION IN THE FEDERAL REGISTER].

The Director of the Federal Register approved the incorporation by reference of a certain publication listed in the AD as of [INSERT DATE 35 DAYS AFTER DATE OF PUBLICATION IN THE FEDERAL REGISTER].

ADDRESSES: You may examine the AD docket on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA-2014-0180;
Discussion

We issued a notice of proposed rulemaking (NPRM) to amend 14 CFR part 39 to add an AD that would apply to Przedsiebiorstwo Doswiadczalno-Produkiejne Model SZD-50-3 "Puchacz" airplanes. The NPRM was published in the Federal Register on March 25, 2014 (79 FR 16248), and proposed to supersede AD 2004-11-10, Amendment 39-13656 (69 FR 31872, June 8, 2004).

The NPRM proposed to correct an unsafe condition for the specified products and was based on mandatory continuing airworthiness information (MCAI) originated by an aviation authority of another country. The MCAI states that:

Several occurrences of airbrake torque tube failure were reported on SZD-50-3 “Puchacz” sailplanes. In all cases, as a result of disruption of the welded joint between torque tube and the lever, the broken torque tube detached from the lever located in the fuselage. The result of subsequent investigations identified fatigue damage, as a consequence of periodical
striking load exceeding the established maximum value, to be a possible failure cause. Additionally, corrosion damage was identified at internal surface of the opened tube.

This condition, if not detected and corrected, would inhibit the function of the airbrake, possibly resulting in reduced control of the sailplane.

Prompted by these findings, Allstar PZL issued Service Bulletin (SB) No. BE-052/SZD-50-3/2003 to provide inspection instructions. CAO of Poland issued AD SP-0052-2003-A to require a one-time inspection of the airbrake torque tube in the area of welded joint in accordance with that SB.

Since that AD was issued, Allstar PZL issued SB No. BE-062/SZD-50-3/2013 to introduce repetitive inspections and accomplishment instructions for reinforced torque tube inspections.

For the reasons described above, this AD supersedes CAO of Poland AD SP-0052-2003-A and requires repetitive inspections of the airbrake torque tube and, depending on findings, replacement with a serviceable part.

The MCAI can be found in the AD docket on the Internet at:


Comments

We gave the public the opportunity to participate in developing this AD. We received no comments on the NPRM or on the determination of the cost to the public.

Conclusion

We reviewed the relevant data and determined that air safety and the public interest require adopting the AD as proposed except for minor editorial changes. We have determined that these minor changes:

- Are consistent with the intent that was proposed in the NPRM (79 FR 16248, March 25, 2014) for correcting the unsafe condition; and
- Do not add any additional burden upon the public than was already proposed in the NPRM (79 FR 16248, March 25, 2014).
Costs of Compliance

We estimate that this AD will affect 5 products of U.S. registry. We also estimate that it would take about 5 work-hours for the annual inspection of sailplanes equipped with the old version torque tube; 1 work-hour for the annual inspection of sailplanes equipped with the new version torque tube; and 5 work-hours for the 1,000-hour annual inspection of sailplanes equipped with the new version torque tube. The average labor rate is $85 per work-hour.

In addition, we estimate that any necessary follow-on actions would take about 5 work-hours and require parts costing $875, for a cost of $1,300 per product. We have no way of determining the number of products that may need these actions.

Authority for This Rulemaking

Title 49 of the United States Code specifies the FAA’s authority to issue rules on aviation safety. Subtitle I, section 106, describes the authority of the FAA Administrator. “Subtitle VII: Aviation Programs,” describes in more detail the scope of the Agency’s authority.

We are issuing this rulemaking under the authority described in “Subtitle VII, Part A, Subpart III, Section 44701: General requirements.” Under that section, Congress charges the FAA with promoting safe flight of civil aircraft in air commerce by prescribing regulations for practices, methods, and procedures the Administrator finds necessary for safety in air commerce. This regulation is within the scope of that authority because it addresses an unsafe condition that is likely to exist or develop on products identified in this rulemaking action.

Regulatory Findings

We determined that this AD will not have federalism implications under Executive Order 13132. This AD will not have a substantial direct effect on the States, on the relationship between the national government and the States, or on the distribution of power and responsibilities among the various levels of government.
For the reasons discussed above, I certify this AD:

(1) Is not a “significant regulatory action” under Executive Order 12866, 
(2) Is not a “significant rule” under the DOT Regulatory Policies and Procedures (44 FR 11034, February 26, 1979),
(3) Will not affect intrastate aviation in Alaska, and
(4) Will not have a significant economic impact, positive or negative, on a substantial number of small entities under the criteria of the Regulatory Flexibility Act.

Examining the AD Docket

You may examine the AD docket on the Internet at http://www.regulations.gov by searching for and locating Docket No. FAA-2014-0180; or in person at the Docket Management Facility between 9 a.m. and 5 p.m., Monday through Friday, except Federal holidays. The AD docket contains the NPRM, the regulatory evaluation, any comments received, and other information. The street address for the Docket Office (telephone (800) 647-5527) is in the ADDRESSES section. Comments will be available in the AD docket shortly after receipt.

List of Subjects in 14 CFR Part 39

Air transportation, Aircraft, Aviation safety, Incorporation by reference, Safety.

Adoption of the Amendment

Accordingly, under the authority delegated to me by the Administrator, the FAA amends 14 CFR part 39 as follows:

PART 39 - AIRWORTHINESS DIRECTIVES

1. The authority citation for part 39 continues to read as follows:

Authority: 49 U.S.C. 106(g), 40113, 44701.

§ 39.13 [Amended]

2. The FAA amends § 39.13 by removing Amendment AD 2004-11-10, Amendment 39-13656 (69 FR 31872, June 8, 2004) and adding the following new AD:
2014-12-08 Przedsiębiorstwo Doswiadczenno-Produkcyjne Szybownictwa “PZL-Bielsko”: Amendment 39-17869; Docket No. FAA-2014-0180; Directorate Identifier 2014-CE-004-AD.

(a) Effective Date

This airworthiness directive (AD) becomes effective [INSERT DATE 35 DAYS AFTER DATE OF PUBLICATION IN THE FEDERAL REGISTER].

(b) Affected ADs

This AD supersedes AD 2004-11-10, Amendment 39-13656 (69 FR 31872; June 8, 2004).

(c) Applicability

This AD applies to Przedsiębiorstwo Doswiadczenno-Produkcyjne Szybownictwa “PZL-Bielsko” Model SZD-50-3 “Puchacz” sailplanes, all serial numbers, certificated in any category.

(d) Subject


(e) Reason

This AD was prompted by continuing airworthiness information (MCAI) originated by an aviation authority of another country to identify and correct an unsafe condition on an aviation product. The MCAI describes the unsafe condition as fatigue damage of the welded joint between the airbrake torque tube and the airbrake control system lever located inside the fuselage. We are issuing this AD to detect and correct fatigue damage of the airbrake torque tube and the airbrake control system lever, which may cause a malfunction of the airbrake, resulting in loss of control of the sailplane.

(f) Actions and Compliance

Unless already done, do the following actions in paragraphs (f)(1) through (f)(6) of this AD:
(1) For sailplanes equipped with the old version torque tube, with or without reinforced corner: Initially within 30 days after [INSERT DATE 35 DAYS AFTER DATE OF PUBLICATION IN THE FEDERAL REGISTER] (the effective date of this AD) and repetitively thereafter at intervals not to exceed every 12 months or 100 hours time-in-service (TIS), whichever occurs first, do a detailed inspection of the airbrake torque tube following the inspection procedures in paragraph (2)(b) in Allstar PZL Glider Sp. z o.o. Service Bulletin No. BE-062/SZD-50-3/2013 “PUCHACZ”, Revision A, dated September 16, 2013.

(2) For sailplanes equipped with the new type torque tube, with reinforced corner: Initially within 30 days after [INSERT DATE 35 DAYS AFTER DATE OF PUBLICATION IN THE FEDERAL REGISTER] (the effective date of this AD) and repetitively thereafter at intervals not to exceed every 12 months or 100 hours TIS, whichever occurs first, visually inspect the welded joint of the airbrake torque tube following the conditions of inspection, first bulleted item of paragraph (2)(a)(2), in Allstar PZL Glider Sp. z o.o. Service Bulletin No. BE-062/SZD-50-3/2013 “PUCHACZ”, Revision A, dated September 16, 2013.

(3) For sailplanes equipped with the new type torque tube, with reinforced corner: During the first 1,000-hour inspection after [INSERT DATE 35 DAYS AFTER DATE OF PUBLICATION IN THE FEDERAL REGISTER] (the effective date of this AD), and then repetitively at each scheduled 1,000-hour inspection, do a detailed inspection of the welded joint of the airbrake torque tube following the inspection procedures in paragraph (2)(b) in Allstar PZL Glider Sp. z o.o. Service Bulletin No. BE-062/SZD-50-3/2013 “PUCHACZ”, Revision A, dated September 16, 2013.

(4) For all sailplanes: If during any inspection required by paragraph (f)(1), (f)(2), or (f)(3) of this AD any damage is found as detailed in paragraph (2)(c) of PZL Glider Sp. z o.o. Service Bulletin No. BE-062/SZD-50-3/2013 “PUCHACZ”, Revision A, dated September 16, 2013, before further flight, replace the airbrake torque tube as described in

(5) For all sailplanes: Replacement of an airbrake torque tube, as required by paragraph (f)(4) of this AD, does not constitute terminating action for inspection requirements of paragraphs (f)(1), (f)(2), and (f)(3) of this AD.

(6) For all sailplanes: Compliance with the requirements of paragraphs (f)(1), (f)(2), or (f)(3) of this AD can be demonstrated by incorporating the applicable required inspections and follow-on corrective actions, as specified in Allstar PZL Glider Sp. z o.o. Service Bulletin No. BE-062/SZD-50-3/2013 “PUCHACZ”, Revision A, dated September 16, 2013, into the approved instructions for continued airworthiness (ICA) of the maintenance program.

(g) Other FAA AD Provisions

The following provisions also apply to this AD:

(1) Alternative Methods of Compliance (AMOCs): The Manager, Standards Office, FAA, has the authority to approve AMOCs for this AD, if requested using the procedures found in 14 CFR 39.19. Send information to ATTN: Jim Rutherford, Aerospace Engineer, FAA, Small Airplane Directorate, 901 Locust, Room 301, Kansas City, Missouri 64106; telephone: (816) 329-4165; fax: (816) 329-4090; email: jim.rutherford@faa.gov. Before using any approved AMOC on any airplane to which the AMOC applies, notify your appropriate principal inspector (PI) in the FAA Flight Standards District Office (FSDO), or lacking a PI, your local FSDO.

(2) Airworthy Product: For any requirement in this AD to obtain corrective actions from a manufacturer or other source, use these actions if they are FAA-approved. Corrective actions are considered FAA-approved if they are approved by the State of Design Authority (or their delegated agent). You are required to assure the product is airworthy before it is returned to service.
(h) Related Information

Refer to MCAI European Aviation Safety Agency (EASA) AD No.: 2014-0015, dated January 14, 2014, for related information. The MCAI can be found in the AD docket on the Internet at: http://www.regulations.gov/#!documentDetail;D=FAA-2014-0180-0002.

(i) Material Incorporated by Reference

(1) The Director of the Federal Register approved the incorporation by reference (IBR) of the service information listed in this paragraph under 5 U.S.C. 552(a) and 1 CFR part 51.

(2) You must use this service information as applicable to do the actions required by this AD, unless the AD specifies otherwise.


(ii) Reserved.


(4) You may view this service information at the FAA, Small Airplane Directorate, 901 Locust, Kansas City, Missouri 64106. For information on the availability of this material at the FAA, call (816) 329-4148.

(5) You may view this service information that is incorporated by reference at the National Archives and Records Administration (NARA). For information on the
availability of this material at NARA, call 202-741-6030, or go to:

Issued in Kansas City, Missouri, on June 6, 2014.

Timothy Smyth,
Acting Manager, Small Airplane Directorate,
Aircraft Certification Service.

[FR Doc. 2014-13839 Filed 06/16/2014 at 8:45 am; Publication Date: 06/17/2014]