



Dream Chaser Rideshare

Ken Shields
Sierra Space

AIRBUS
Technical Session Sponsor





Company Overview

A DIFFERENTIATED SPACE LEADER

A top-tier vertically integrated space-tech company revolutionizing the space industry and disrupting terrestrial industrial markets.



EARTHSPACE SYSTEMS

Revolutionary spaceplane with global commercial runway access for spaceflight operations

First business-ready commercial space station to harness the foundational tech of microgravity



SPACE APPLICATIONS

Differentiated technology for critical satellite and spacecraft systems

Next generation defense prime in classified space

SIERRA
SPACE



BY THE NUMBERS

\$4.3B

Active Contracts with
>\$400M in National
Security

70

Active Customers

\$1.7B

Series A & B
(\$5.3B Valuation)

75

Patents, Pending
Applications and Trade
Secrets

2,000 | 425

Team Members
| Advanced Degrees

950,000

Square Feet of
Terrestrial Infrastructure

2

Mission Control Rooms

28

Acre Rocket Engine
Test Site

2

Centers in Support of
National Security

500

Space Missions Enabled
by Sierra Space

49

Interplanetary Missions

6 | 30

Years Developing
Dream Chaser® | Years
Enabling Space Missions

Sierra Space US Presence

Infrastructure in place to scale



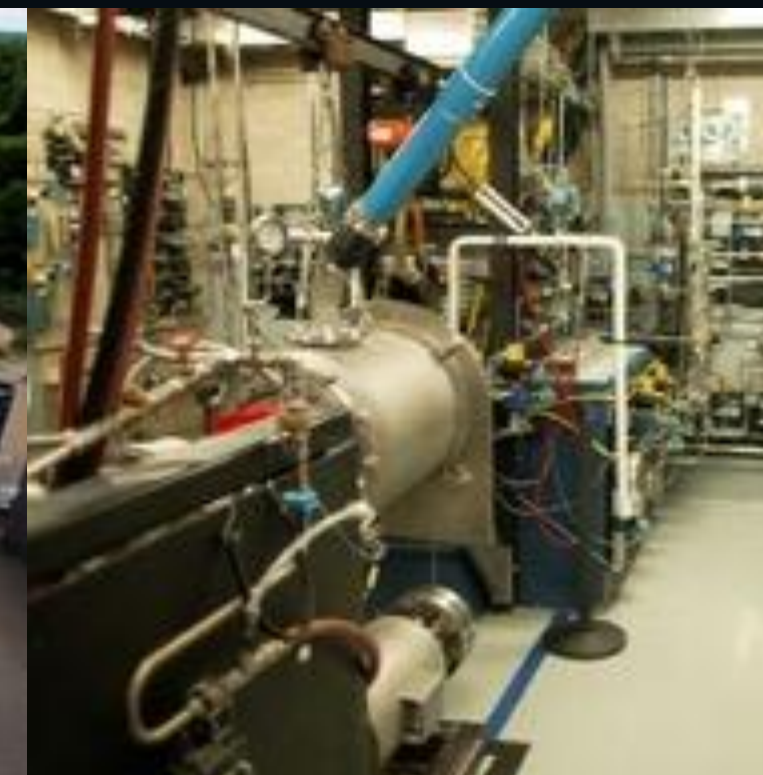
Mission Control
Louisville, Colorado



Vehicle Development
Louisville, Colorado



Rocket Engine Development & Test
North Freedom, WI



Electromechanical
Durham, NC



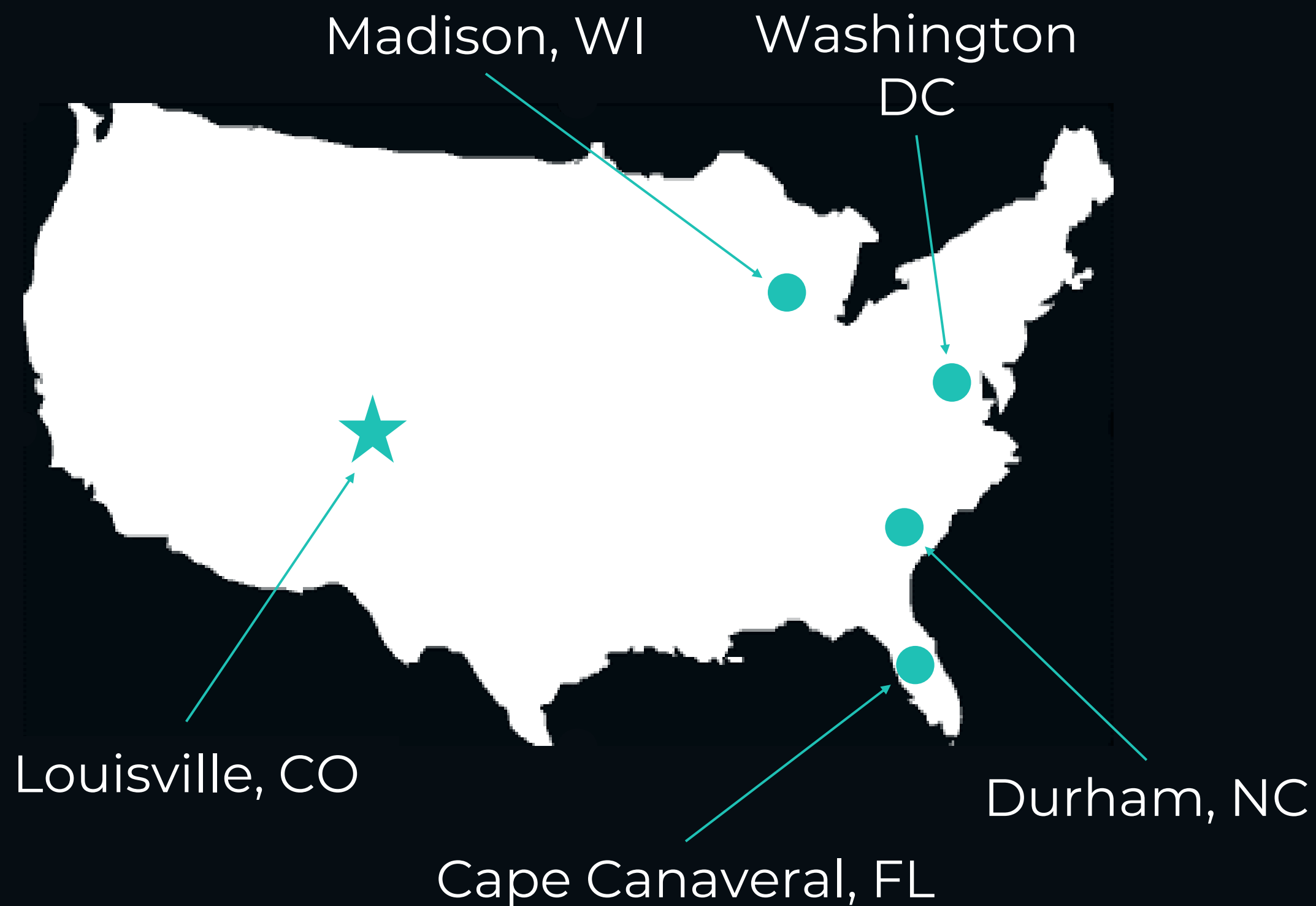
LIFE Support & Propulsion
Madison, WI



Power & Avionics
Louisville, Colorado



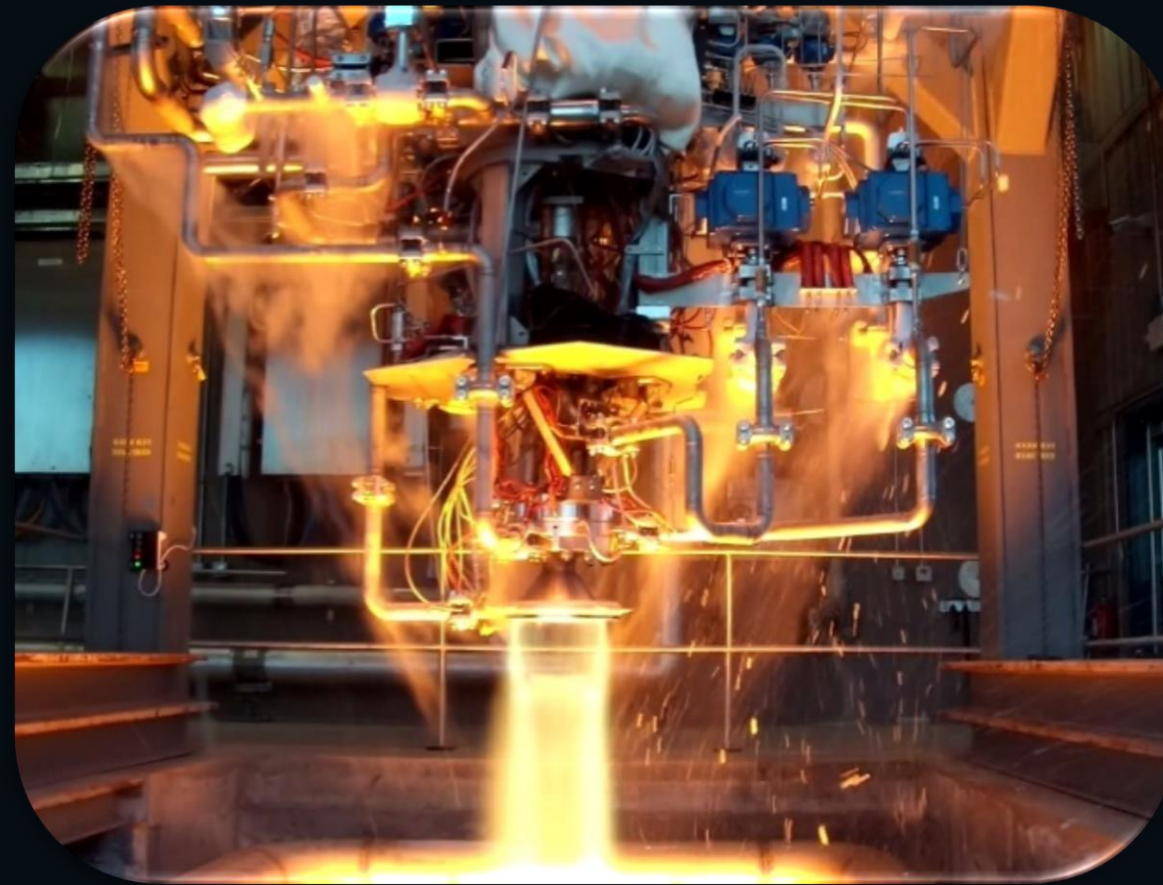
Launch Operations
KSC, Florida



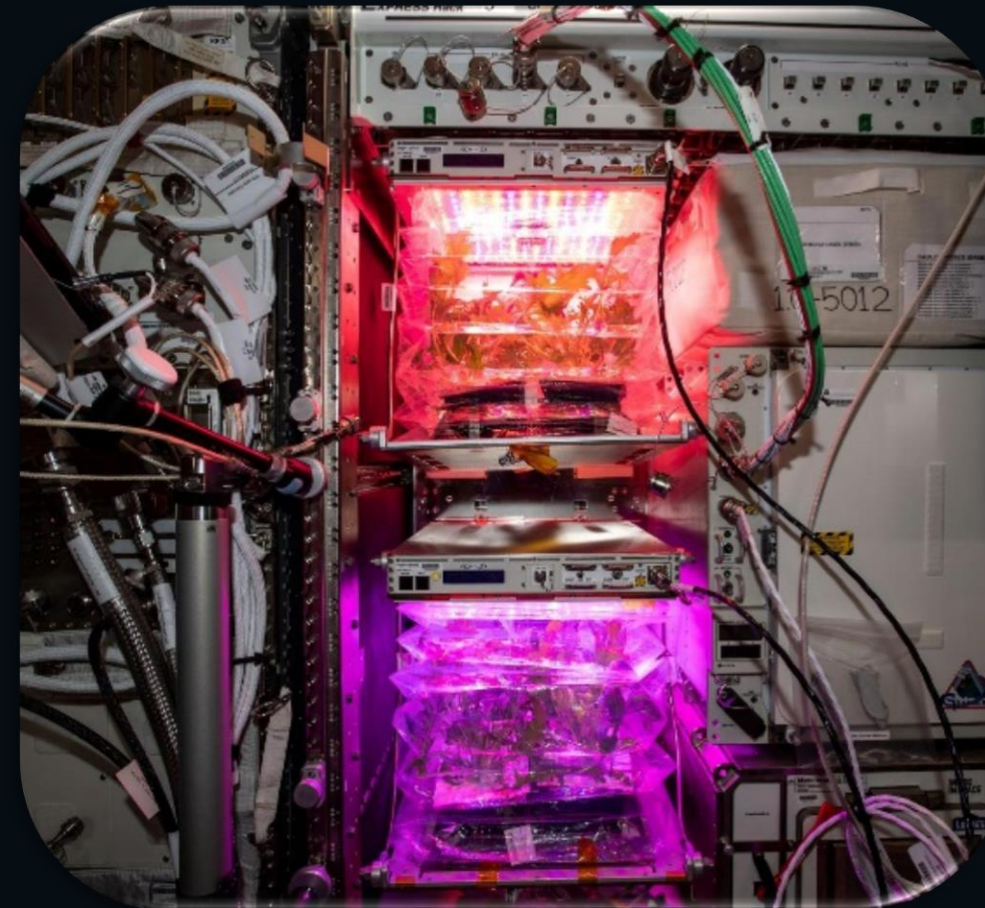


Subsystems and Components

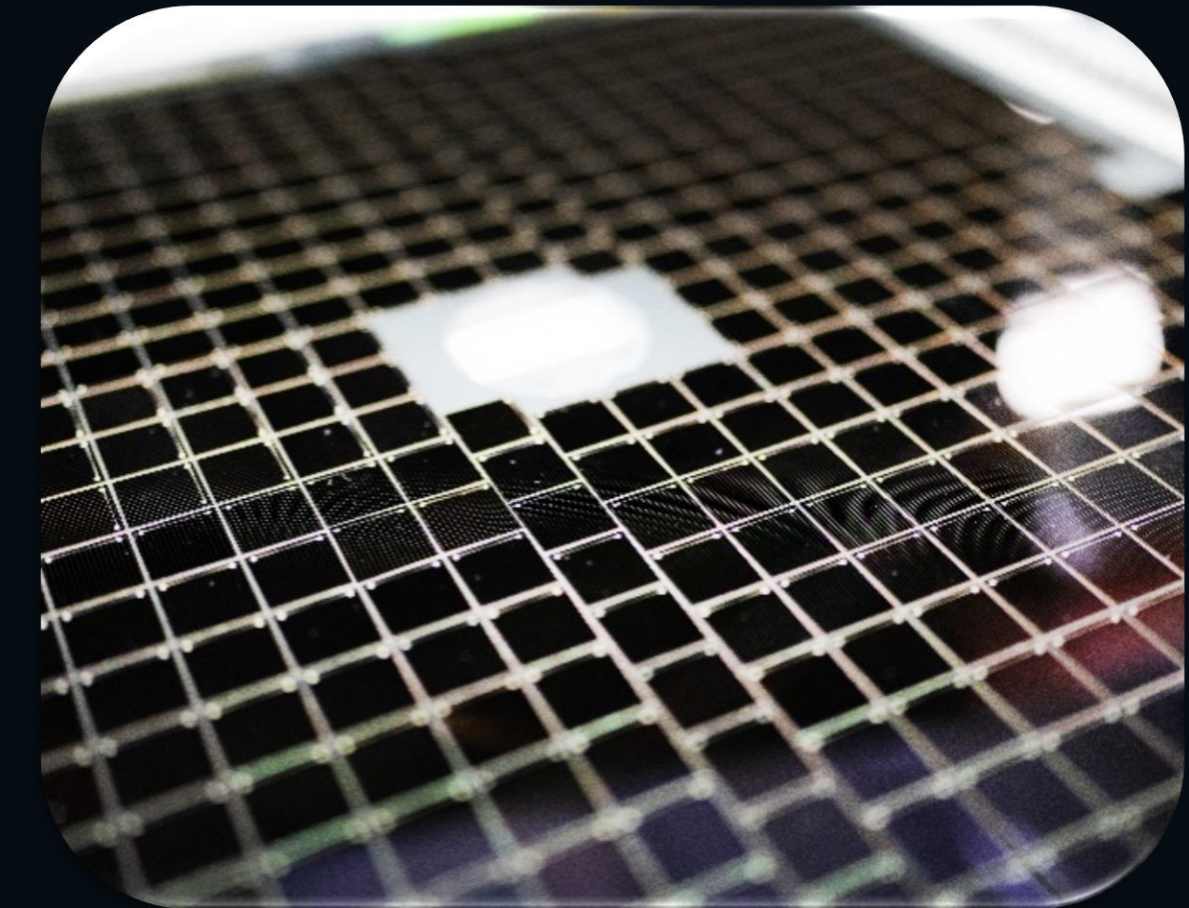
Propulsion Systems



Environmental Control Systems



Power Systems & Mechanisms

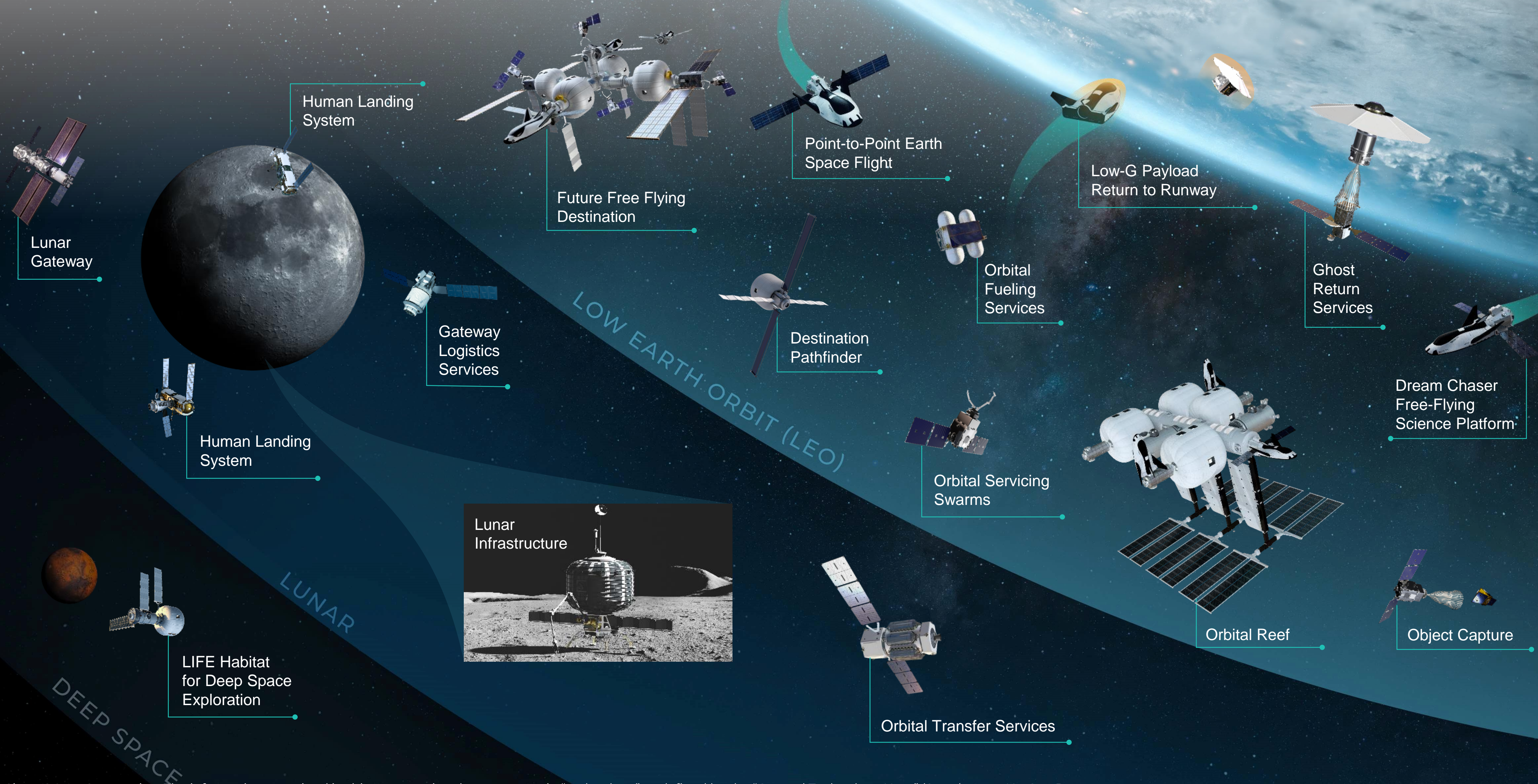


Powering Aerospace Customers





UNIQUELY POSITIONED TO BUILD THE FUTURE SPACE INFRASTRUCTURE



DREAM CHASER®

REVOLUTIONIZING SPACE TRAVEL



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Dream Chaser System Overview

- \$3B** Active NASA Contract for Cargo to ISS
- 15+** Missions per Spaceplane – Highly Reusable, Runway Capable
- 6+** Tons Cargo Capacity
- 2024** First Cargo Mission to ISS later in 2024
- 2nd** Spaceplane Under Construction
Provides Numerous Mission Flexibility

SIERRA
SPACE



Pre-Launch

Weeks

- Unpressurized payload installation on Shooting Star
- Pressurized payload installation including late load within 24 hours of launch
- Launch vehicle fairing encapsulation with access to cargo available
- Power, command, and data available if needed



Launch Prep

One Day

- Spacecraft power on
- Go/No go for launch
- Spacecraft to internal power



Post Insertion

Hours

- Solar array deploy
- Dream Chaser wing deploy
- Critical systems checkout



ISS Operations

*Days to Months
(Mission dependent)*

- Mission activities

Payload Deploy and Dream Chaser Deorbit

Hours

- Optional Dream Chaser reboost for payload deploy
- Deorbit burn, Shooting Star separation



Post-Landing

Hours to Days

- Dream Chaser recovery and power down
- Pressurized payload and time-sensitive cargo retrieval within hours of landing



Shooting Star and Unpressurized Payload Disposal

Minutes



Entry, Descent, and Landing

Minutes





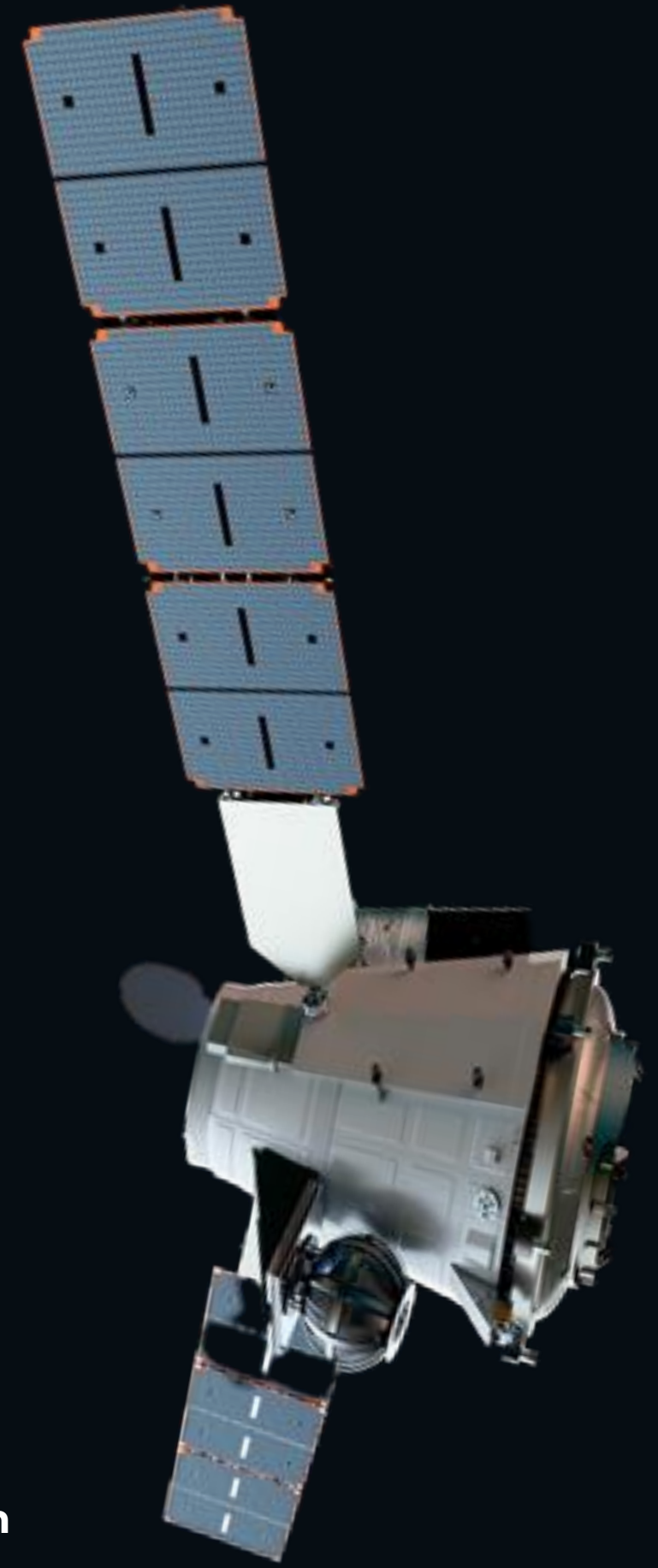
Rideshare Overview

ConOps

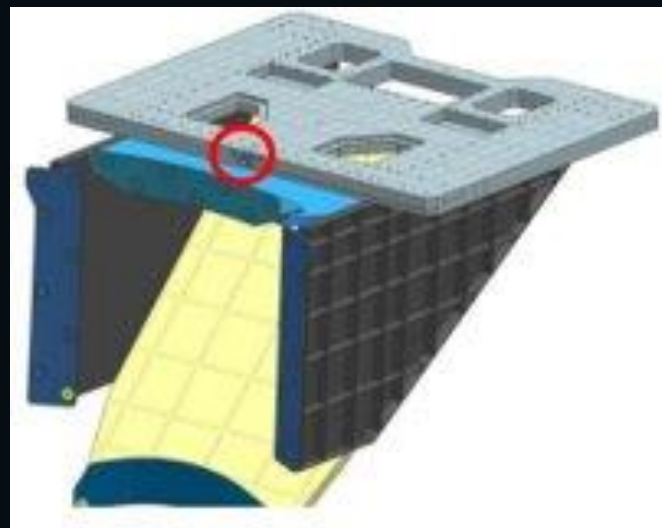
- **ISS Rideshare:**
 - Deployment may happen prior to ISS (first 3 days of mission) or post-ISS (after 30-75 days berthed)
 - Altitude from 330 – 450 km, inclination 51.6°
- **For non-ISS missions:**
 - Deployment specifics are dependent on the individual mission
 - Altitude up to 450 km, inclination 28.5 - 51.6°

Interfaces

- Deployable mechanism to be built by payload-provider to interface with one of the mounting options
 - Optionally, deployment mechanism can be provided by Sierra Space
- Dream Chaser team will perform deployment activity (e.g. commanding, monitoring)
- Flush mounted multi-use interface plate for hosted payloads



Flush Mount Interface Plate



QwkSep 15 Low-Shock Separation System



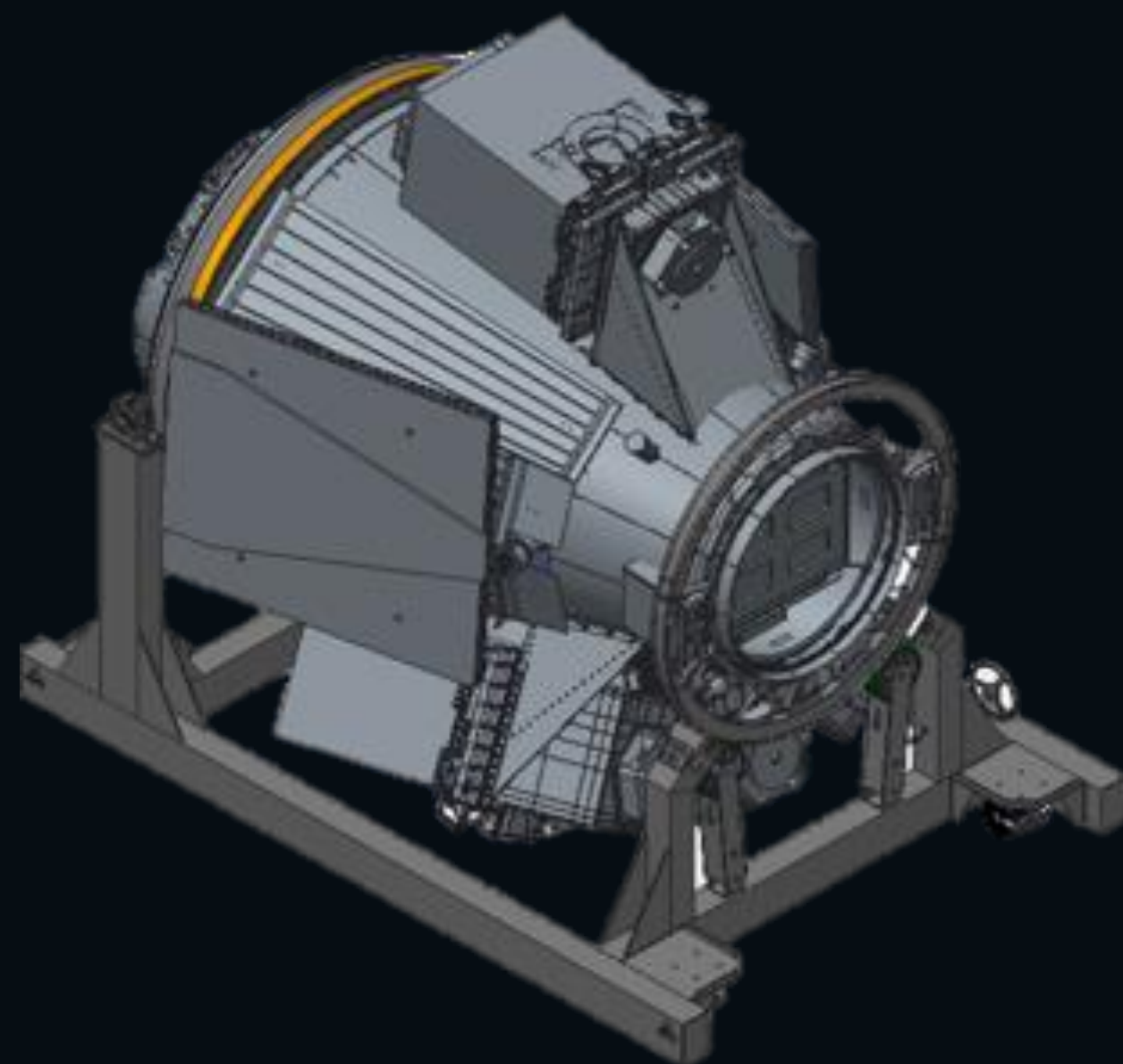
QwkSep 24 Low-Shock Separation System





Payload Integration ConOps

- Payloads are delivered to KSC at ~1 month before launch for installation at the Space Station Processing Facility (SSPF)
- Access for Remove Before Flight items is available within 7 days of launch, once within the fairing and stacked with the Launch Vehicle
- Once installed, the following resources are available:

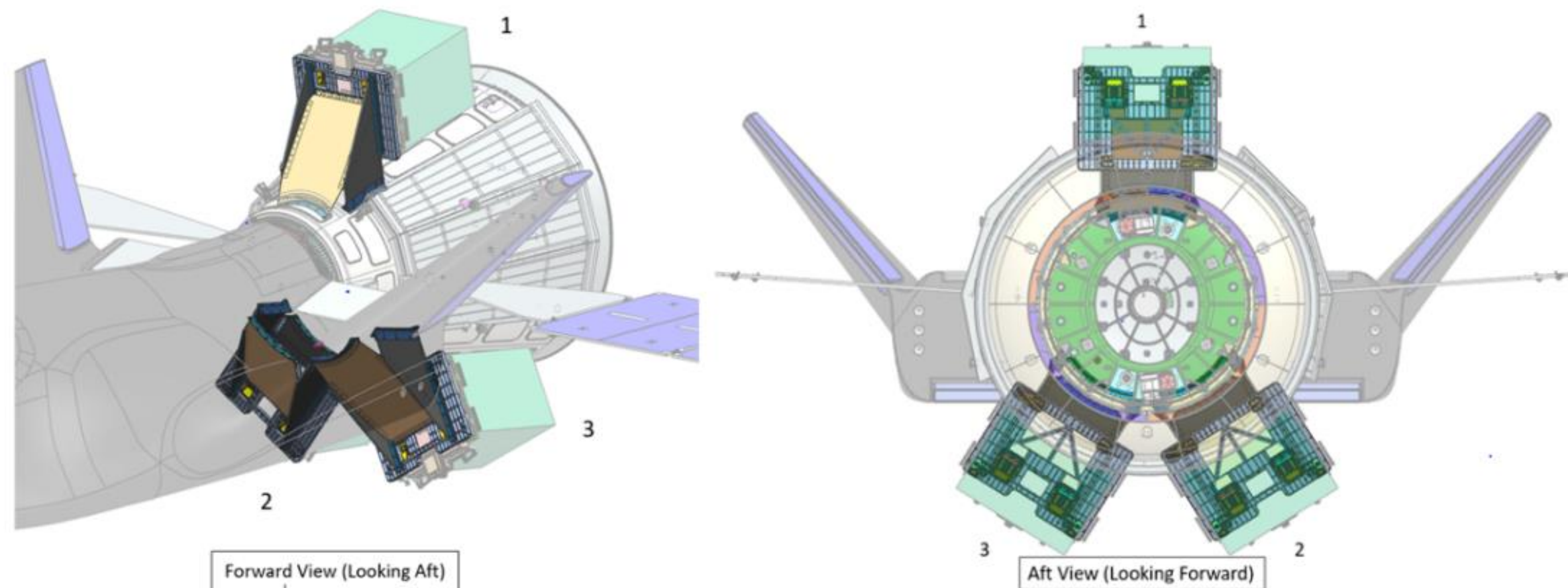


Resource	Nominal	During Transportation Between Facilities
Power	300 W 120 Vdc; Provided via ground power resource to support operations as needed	None
Data	When powered, health and status data are available once per day; limited to 0.1 Hz post-encapsulation	None
Commanding	Provided if needed when the spacecraft is powered. Once at the launch pad, no commanding available.	None
Temperature and Humidity Control	Pre-VIF: 60 °F (15.56 °C) to 80 °F (26.67 °C) VIF and after: 50 °F (10°C) to 80 °F (26.67 °C) Pre-VIF: 25% to 75% relative humidity VIF and after: 0% to 75% relative humidity	50 °F (10.0 °C) to 89 °F (31.6 °C) 0% to 75% relative humidity
Air Purge	Provided if needed pre-encapsulation and then continuous 5K class post-encapsulation. At L-4 hours, switches to 5k class GN2.	Varies between 100k purge to visibly clean





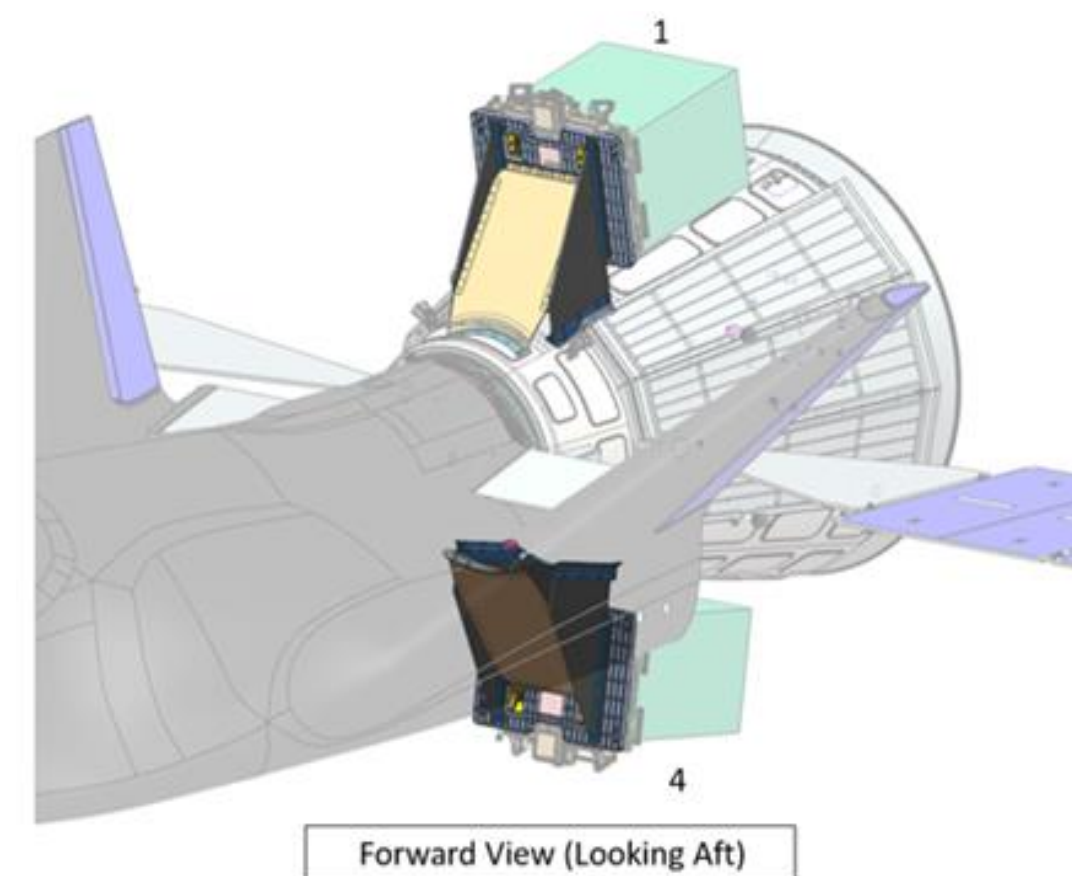
Mounting Locations



Configuration 1

The CM can be arranged to support several different configurations of payloads using four locations. Locations 1, 2, and 3 are shown left. Locations 1 and 4 are shown below.

A payload complement may consist of up to **three unpressurized payloads**. The maximum total unpressurized cargo complement is **3,307 lbs. (1,500 kg)**.



Configuration 2





Orbit Options

Orbit Altitude Profile	Inclination	Expected Duration
333 km (LV Separation)	51.65°	3 to 12 hours
333 to 420 km	51.65°	Up to 2 days
420 km (ISS)	51.65°	Up to 75 days
420 to 450 km*	51.65°	Up to 1 day
420 km (Deorbit)	51.65°	Up to 1 day
<i>*Optional for deployments, circular and elliptical orbits available</i>		





Interfaces

Interface Type	Volume Envelope (Inches)	Mass Capability (Kg)	Power (Watts)	Supported Satellite Class
Flush Mount	83 x 44 x 58 x 41 (L x W x H1 x H2)	500	100 W average power 333 W peak power for up to 2 cumulative hours	Up to ESPA Grande
90° Standoff	48 x 56 x 44 x 38 (L x W x H1 x H2)	500		Up to ESPA Grande
QwkSep® 15	83 x 44 x 58 x 41 (L x W x H1 x H2)	181		ESPA Class
QwkSep® 24	83 x 44 x 58 x 41 (L x W x H1 x H2)	300		ESPA Grande Class
Note: Separation System volume envelopes include separation ring with a 2” height				





Example Timeline

Event	Date
Integration Kickoff	L-20 months
Vehicle Baseline Review (VBR)	L-18 months
Payload Data Delivery	L-16 months
Draft ICD	L-14 months
Mission Integration Review (MIR)	L-12 months
Final ICD	L-10 months
Unpressurized Integration Review (UIR)	L-10 months
Payload Data Delivery	L-8 months
Flight Operations Review (FOR)	L-6 months
Cargo Integration Review (CIR)	L-3 months
Payload Installation	L-60-40 days
Flight Readiness Review (FRR)	L-4 days
Launch Vehicle Readiness Review (LRR)	L-2 days
Post-Mission Review	End of Mission (EOM) +30 days



Thank you CASIS, AIAA, NASA and Sponsors



Please reach out!

Kenneth.shields@sierraspace.com

Or

sales@sierraspace.com

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